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# Minimum-drag Satellite Geometries in VLEO



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VLEO: Orbits below 450km altitude. The atmosphere has a significant effect on spacecraft. Most common surfaces become contaminated by adsorbed atomic oxygen.

Reflection is generally quasi-specular or diffuse, rather than specular, which affects the aerodynamic characteristics [1].

The choice of atmospheric model leads to uncertainties, based on factors such as geomagnetic and solar activity.

The uncertainty in GSIs and the atmospheric model leads to a lower limit on the uncertainty of drag coefficients.



The figure above shows some shape optimisation that has already been investigated [1]. A common characteristic of the papers concerning this subject is a lack of continuity between configurations, with discrete shapes being investigated independently.



# Background:

- Gas-surface
- interaction (GSI)
- Atmospheric models

• Drag modelling

• Shape optimisation

Panel methods are the fastest way to calculate aerodynamic drag. In-house program which does this: ADBSat [2].



Top: Specular reflection Bottom: Quasi-specular (yellow), and diffuse (grey) reflection



DSMC grid visualisation A numerical method for modelling gas flows. Models many real particles as one DSMC particle.

Calculates intermolecular and surface collisions probabilistically to solve the Boltzmann equations.

Choice of program: OpenFoam. Has been validated against analytical and discrete methods [3] Usage: to validate and examine the limits of ADBSat.

Pros Convex geometries Multiple reflections Fewer assumptions

Cons Computationally expensive Time intensive



### **Direct Simulation** Monte Carlo (DSMC): • Validation of panel methods

#### Many choices of optimisation algorithm:



Problem space is represented by some function f(x), where  $\mathbf{x}$  is a vector of n optimisation variables. Challenges in finding the optimal solution include:



## Computational **Optimisation:**

- Optimisation algorithm choices
- 3D optimisation problems



### Particle Swarm Optimisation Hybrid Methods

No Free Lunch theorem: No one algorithm is better than any other for all problems. To pick the best one, you need an understanding of the underlying problem space and each specific search method [4].

- Developing an accurate mathematical representation of f(x)
- Handling many degrees of freedom in a reasonable computational time
- Identifying symmetries and other simplifying factors
- Handling constraints
- Devising a goodness-of-fit measure.



#### optimum point of a 3D solution space



Satellite shape alteration

Validate ADBSat for complex geometries

Formulate a mathematical description of shape optimisation problem

Write an optimisation routine which attempts to find the best solution to the problem

Establish a method for finding lowest-drag configurations subject to constraints

Use DSMC and analytical results to find limits of ADBsat, i.e. cases where it significantly deviates from solution.

Represent the problem and its solution in mathematical terms using set theory. Determine which way to best search the solution set for the optimum solution.

Search for the optimum low-drag solution globally, subject to practical constraints. Integrate the geometry optimisation with ADBSat to perform the drag modelling calculations.

Biggest challenges: Many degrees of freedom, computational expense, handling constraints, establishing a goodness-of-fit measure.

## Preliminary project plan:

• Goal: to identify new optimised aerodynamic geometries.



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